

$$\Sigma \vec{F} = \frac{d\vec{p}}{dt} \quad \Sigma \vec{M} = \frac{d\vec{h}}{dt}$$

- momentum - angular momentum

$$\vec{R} = \vec{R}_G + \vec{r}$$

$\vec{v} \equiv$ velocity of dm

$$= \frac{d\vec{R}_G}{dt} + \frac{d\vec{r}}{dt} + \vec{\omega} \times \vec{r}$$

\vec{r} is a fixed vector

$$d\vec{H}_O = \vec{R} \times (dm \cdot \vec{v}) = (\vec{R}_G + \vec{r}) \times (\vec{v}_G + \vec{\omega} \times \vec{r})$$

$$= \vec{R}_G \times \vec{v}_G \cdot dm + \vec{R}_G \times (\vec{\omega} \times \vec{r}) dm + \vec{r} \times \vec{v}_G dm + \vec{r} \times (\vec{\omega} \times \vec{r}) dm$$

$$\vec{H}_O = \vec{R}_G \times \vec{v}_G \int_m dm + \vec{R}_G \times \vec{\omega} \times \int_m \vec{r} dm - \vec{v}_G \times \int_m \vec{r} dm + \int_m \vec{r} \times (\vec{\omega} \times \vec{r}) dm$$

if point G is the center of mass $\Rightarrow \int_m \vec{r} dm = 0$

also $\int_m dm = M$ - total mass of object

$$\vec{H}_O = \vec{R}_G \times \vec{v}_G M + \underbrace{\int_m \vec{r} \times (\vec{\omega} \times \vec{r}) dm}_{\text{angular momentum about G}}$$

$$\int_m \vec{r} \times (\vec{\omega} \times \vec{r}) dm = \mathbb{I} \vec{\omega}$$

mass distribution inertial tensor

$$\vec{H}_O = \vec{R}_G \times \vec{v}_G M + \vec{H}_G$$

angular momentum about G

$$\vec{H}_G = \int_m \vec{r} \times (\vec{\omega} \times \vec{r}) dm = \omega \int_m r^2 dm - \int_m \vec{r} (\vec{r} \cdot \vec{\omega}) dm$$

note: $\vec{A} \times (\vec{B} \times \vec{C}) = \vec{B}(\vec{A} \cdot \vec{C}) - \vec{C}(\vec{A} \cdot \vec{B})$

$\vec{r} \times (\vec{\omega} \times \vec{r}) = \omega \cdot r^2 - \vec{r}(\vec{r} \cdot \vec{\omega})$

in cartesian coordinates: $\vec{\omega} = \omega_x \hat{i} + \omega_y \hat{j} + \omega_z \hat{k}$
 $\vec{r} = x \hat{i} + y \hat{j} + z \hat{k}$

$$\vec{r} \cdot (\vec{r} \cdot \vec{\omega}) = \vec{r} (x\omega_x + y\omega_y + z\omega_z)$$

expanding integrals

$$H_G = (\omega_x \hat{i} + \omega_y \hat{j} + \omega_z \hat{k}) \int_m (x^2 + y^2 + z^2) dm - \int_m (x\omega_x + y\omega_y + z\omega_z) (x\hat{i} + y\hat{j} + z\hat{k}) dm$$

combining like terms

$$H_G = \int_m \left\{ \begin{aligned} &[(x^2 + y^2 + z^2)\omega_x - (x^2\omega_x + xy\omega_y + xz\omega_z)] \hat{i} + \\ &[(x^2 + y^2 + z^2)\omega_y - (yx\omega_x + y^2\omega_y + yz\omega_z)] \hat{j} + \\ &[(x^2 + y^2 + z^2)\omega_z - (zx\omega_x + zy\omega_y + z^2\omega_z)] \hat{k} \end{aligned} \right\} dm$$

$$= \int_m \left\{ \begin{aligned} &[(y^2 + z^2)\omega_x - (xy)\omega_y - (xz)\omega_z] \hat{i} + \\ &[-(xy)\omega_x + (x^2 + z^2)\omega_y - (yz)\omega_z] \hat{j} + \\ &[-(zx)\omega_x - (zy)\omega_y + (x^2 + y^2)\omega_z] \hat{k} \end{aligned} \right\} dm$$

$$I_{xx} = \int_m (y^2 + z^2) dm$$

moments of inertia

- represent mass distribution around an axis

$$I_{xy} = - \int_m xy dm$$

products of inertia

- represent symmetry of mass distribution in the plane

$$H_G = \begin{bmatrix} I_{xx} & I_{xy} & I_{xz} \\ I_{xy} & I_{yy} & I_{yz} \\ I_{xz} & I_{yz} & I_{zz} \end{bmatrix} \begin{bmatrix} \omega_x \\ \omega_y \\ \omega_z \end{bmatrix}$$

symmetric tensor \mathbb{I}

fix xyz in the rigid body $\therefore \mathbb{I} = \text{const}$

$$\bar{H}_G = \mathbb{I} \bar{\omega} \quad (\bar{\omega} \neq \bar{H} \text{ are functions of time})$$

suppose M_G (moment about CM) is known

$$\bar{M}_G = \underbrace{\frac{d\bar{H}_G}{dt}}_{\text{inertial reference}} = \underbrace{\left(\frac{d\bar{H}_G}{dt} \right)_{\text{rel}}}_{\text{relative reference}} + \bar{\omega} \times \bar{H}_G$$

in a symmetric body

$$\mathbb{I} = \begin{bmatrix} I_{xx} & 0 & 0 \\ 0 & I_{yy} & 0 \\ 0 & 0 & I_{zz} \end{bmatrix}$$

$$\bar{H}_G = I_{xx} \omega_x \hat{i} + I_{yy} \omega_y \hat{j} + I_{zz} \omega_z \hat{k}$$

$$\bar{M}_G = M_x \hat{i} + M_y \hat{j} + M_z \hat{k}$$

$$\bar{\omega} \times \bar{H}_0 = \underbrace{\begin{vmatrix} \hat{i} & \hat{j} & \hat{k} \\ \omega_x & \omega_y & \omega_z \\ I_{xx}\omega_x & I_{yy}\omega_y & I_{zz}\omega_z \end{vmatrix}}_{\text{determinant}} = (I_{zz}\omega_z\omega_y - I_{yy}\omega_y\omega_z)\hat{i} - (I_{zz}\omega_z\omega_x - I_{xx}\omega_x\omega_z)\hat{j} + (I_{yy}\omega_y\omega_x - I_{xx}\omega_x\omega_y)\hat{k}$$

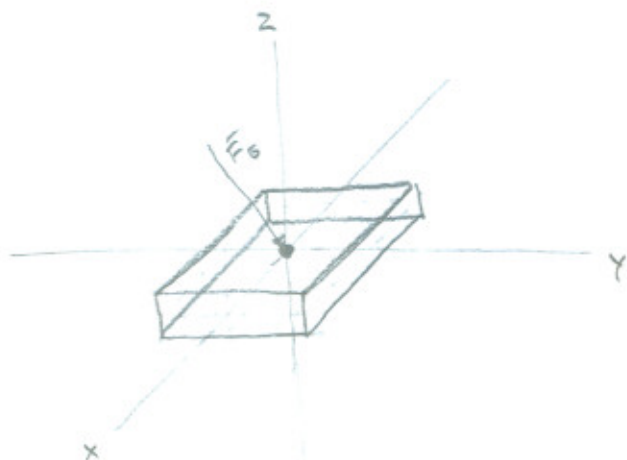
$$\left(\frac{dH_0}{dt}\right)_{rel} = I_{xx}\dot{\omega}_x\hat{i} + I_{yy}\dot{\omega}_y\hat{j} + I_{zz}\dot{\omega}_z\hat{k}$$

sub back into $M_0 = \left(\frac{dH_0}{dt}\right)_{rel} + \bar{\omega} \times \bar{H}_0$

$$\begin{aligned} \hat{i}: M_x &= I_{xx}\dot{\omega}_x - (I_{yy} - I_{zz})\omega_y\omega_z \\ \hat{j}: M_y &= I_{yy}\dot{\omega}_y - (I_{zz} - I_{xx})\omega_z\omega_x \\ \hat{k}: M_z &= I_{zz}\dot{\omega}_z - (I_{xx} - I_{yy})\omega_x\omega_y \end{aligned} \left. \vphantom{\begin{aligned} \hat{i}: M_x &= I_{xx}\dot{\omega}_x - (I_{yy} - I_{zz})\omega_y\omega_z \\ \hat{j}: M_y &= I_{yy}\dot{\omega}_y - (I_{zz} - I_{xx})\omega_z\omega_x \\ \hat{k}: M_z &= I_{zz}\dot{\omega}_z - (I_{xx} - I_{yy})\omega_x\omega_y \end{aligned}} \right\} \text{non-linear Euler's Equations}$$

note: in 2-D $\bar{\omega}$ & \bar{H} are colinear
in 3-D $\bar{\omega}$ & \bar{H} are not necessarily colinear

Book Problem 3



$$I_z > I_y > I_x$$

torque free motion $\bar{M} = 0$

$$\dot{\omega}_x = \underbrace{\left(\frac{I_y - I_z}{I_x} \right)}_{(A)} \omega_y \omega_z$$

A is negative

$$\dot{\omega}_y = \underbrace{\left(\frac{I_z - I_x}{I_y} \right)}_{(B)} \omega_x \omega_z$$

B is positive

C is negative

$$\dot{\omega}_z = \underbrace{\left(\frac{I_x - I_y}{I_z} \right)}_{(C)} \omega_x \omega_y$$

rotation about z-axis

initial conditions

$$\omega_z = \Omega$$

$$\omega_x = \epsilon$$

$$\omega_y = \delta$$

small errors

$$\dot{\omega}_x = A \Omega \delta$$

(drop terms of order ϵ^2, δ^2)

$$\dot{\omega}_y = B \Omega \delta$$

$$\dot{\omega}_z = C \epsilon \delta \sim 0$$

$$\dot{\epsilon} = (A \Omega) \delta$$

$$\ddot{\epsilon} = (A \Omega) \dot{\delta} = AB \Omega^2 \epsilon$$

$$\ddot{\epsilon} - \underbrace{(AB \Omega^2)}_{\text{negative value}} \epsilon = 0$$

$$\dot{\delta} = (B \Omega) \epsilon$$

$$\dot{\delta} = (B \Omega) \dot{\epsilon} = AB \Omega^2 \delta$$

$$\ddot{\delta} + k^2 \delta = 0$$

$$\epsilon = D \cos kt + E \sin kt$$

$$\delta = \dots$$

stable for this case:

- initially valid solution

ϵ, δ doesn't grow



rotation about x-axis

$$\omega_z = \epsilon \quad \dot{\omega}_x = A\delta\Omega$$

$$\omega_y = \delta \quad \dot{\omega}_y = B\epsilon\Omega$$

$$\omega_x = \Omega \quad \dot{\omega}_z = \epsilon\delta\Omega \sim 0$$

$$\dot{\delta} = B\epsilon\Omega \quad \ddot{\delta} = B\Omega\dot{\epsilon} = AB\Omega^2\delta$$

$$\dot{\epsilon} = A\delta\Omega \quad \ddot{\epsilon} = A\Omega\dot{\delta} = AB\Omega^2\epsilon$$

$$\ddot{\delta} - AB\Omega^2\delta = 0$$

$$\ddot{\delta} + k^2\delta = 0$$

$$\delta = D \cdot \cos kt + E \cdot \sin kt$$

$$\epsilon = \dots$$

- similar stable condition such as in 2-axis rotation

rotation about y-axis

$$\omega_x = \epsilon \quad \dot{\omega}_x = A\delta\Omega$$

$$\omega_y = \Omega \quad \dot{\omega}_y = B\epsilon\delta \sim 0$$

$$\omega_z = \delta \quad \dot{\omega}_z = C\epsilon\Omega$$

$$\dot{\epsilon} = A\delta\Omega \quad \ddot{\epsilon} = A\Omega\dot{\delta} = AC\Omega^2\epsilon$$

$$\dot{\delta} = C\epsilon\Omega$$

$$\ddot{\epsilon} - \overbrace{AC\Omega^2}^{\text{positive}} \epsilon = 0$$

$$\ddot{\epsilon} - k^2\epsilon = 0$$

$$\epsilon = D \cdot e^{kt} + E \cdot e^{-kt}$$

initially ϵ, δ have an exponential increase, decrease



to solve for full motion in time the non-linear Eulers equations need to be solve simultaneously, probably numerically.